

MANDATORY SERVICE BULLETIN MSB 40NG-018/1 SUPERSEDES MSB 40NG-018

I <u>TECHNICAL DETAILS</u>

I.1 Category

Type:

Mandatory.

I.2 Airplanes affected

Serial numbers:

DA 40 NG

DA 40 NG

40.N001 through 40.N084 Airplanes converted to DA 40 NG with WI-OSB 40-080 up to Rev. 6.

NOTE Airplanes originally modified with RSB 40NG-011 are also affected.

I.3 Date of effectivity

22-Nov-2012

I.4 <u>Time of Compliance</u>

Action 1: Within the next 50 flight hours or at next maintenance action whichever occurs first and every 50 flight hours thereafter until Action 2 is carried out.

Action 2: Within the next 200 flight hours or 12 month from the date of effectivity which ever occurs first.

I.5 Subject

Exchange of charged air tubing.

ATA-Code: 81-00

I.6 <u>Reason</u>

Leakages of the charge air tubing due to improper installation and maintenance as well as chafing of aluminum tubing on the firewall because of small clearances making the installation critical required a new design providing more clearance. This service bulletin describes the retrofit installation of the new design.

I.7 Concurrent Documents

None.

I.8 <u>Approval</u>

The technical information or instructions contained in this document relate to the Design Change Advisory No. MÄM 40-630, which has been approved under the authority of EASA Design Organization Approval ref. EASA.21J.052.



The technical content of this document has been approved under the authority of DOA ref. EASA.21J.052.

I.9 Accomplishments / Instructions

Action 1: Inspection

Inspect of the charge air tubing from the turbo charger to the intercooler: Look especially for marks of chafing. No chafing marks allowed.

If chafing marks are detected either replace parts per AMM or comply with Action 2.

Action 2: Installation of new charge air tubing:

Comply with WI-MSB 40NG-018, latest effective issue.

I.10 Mass (Weight) and CG

The change in mass and CG is negligible.

II PLANNING INFORMATION

II.1 Material and Availability

See WI-MSB 40NG-018, latest effective issue.

II.2 Special Tools

None.

II.3 Labour Effort

Approx. 1.5 hours.

II.4 Credit

For airplanes in warranty:

- Action1: None for inspection only. In case of findings, replacement parts may be ordered free of charge.
- Action 2: Material as listed in WI-MSB 40-018 and 1.5 hours of labour.

II.5 <u>Reference Documents</u>

DA 40 NG Airplane Maintenance Manual, Doc. No. 6.02.15, latest effective issue.



III <u>REMARKS</u>

- 1. All work must be done by a certified aircraft service station or a certified aircraft maintenance mechanic.
- 2. All work, particular those that are not especially described in this Service Bulletin, must be done in accordance with the referenced Maintenance Manual.
- 3. Completion of all work must be recorded in the log book.
- 4. If material and/or labor hours are subject to be credited through Diamond Aircraft Industries, the Service Bulletin must be carried out by an authorized Diamond Service Center and the Warranty Application incl. Work Report must be sent not later than 30 days after the end of time of compliance.
- 5. In case of doubt contact Diamond Aircraft Industries GmbH.



Diamond Aircraft Industries GmbH N. A. Otto-Straße 5 A-2700 Wiener Neustadt Austria DAI MSB 40NG-018/1 Page 4 of 4 26-Nov-2012

EXECUTION REPORT TO SERVICE BULLETIN MSB 40NG-018/1

AIRPLANE DATA			
Airplane Serial Number:			_
Airplane Registration:			_
Airplane Operator:			_
Hours of operation of airplane:			_
No. of landings:			_
Hours of operation-engine			_
Typical operation of airplane:	private, club	o, training, other	

Date, Name, Sign

Please fax the completed form to Fax No. +43-2622-26700-1369 or e-mail to airworthiness@diamond-air.at



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WORK INSTRUCTION

WI-MSB 40NG-018

I GENERAL INFORMATION

I.1 Subject

Replacement of charge air tubes.

I.2 <u>Reference Documents</u>

DA 40 NG Airplane Maintenance Manual, Doc. No. 6.02.15, latest effective issue. DAI SI 40NG-007 Installation and tightening torque of worm drive clamps.

I.3 Remarks

- a) All work must be done by a certified aircraft service station or a certified aircraft maintenance mechanic.
- b) All work, in particular if not described in this work instruction, must be done in accordance with the referenced maintenance manual.
- c) For conversion factors between SI units and US/Imperial units refer to AMM Chapter 02.
- d) In case of doubt, contact Diamond Aircraft Industries GmbH.

II DRAWINGS, SPECIAL TOOLS & MATERIALS

II.1 Drawings

None.

II.2 Special Tools

None.



II.3 <u>Material</u>

If RSB 40NG-011 or MÄM 40-506 or MÄM 40-502 was carried out, use the material below:

Quantity	Part Number	Description	
1	D44-8122-20-00_2 or D44-8122-20-00_4	Turbo Connector Assy	
1	D44-8126-00-05_2	Charge Air Tube	
1	DIN 3771 46.2x3.55 MVQ 70	O-Ring	
8	TORRO_60-80_12_W1	Worm drive Clamps	

If RSB 40NG-011 or MÄM 40-506 or MÄM 40-502 was **NOT** carried out, use the material below:

Quantity	Part Number	Description		
1	D44-8126-00-08	Charged Air Hose		
1	D44-8122-00-10_1	Elbow 45° (Hose)		
8	TORRO_60-80_12_W1	Worm Drive Clamps		
1	DIN 933-M8x16-YZP	Hexagon head screw		
1	D44-8122-20-00_2 or D44-8122-20-00_4	Turbo Connector Assy		
1	D44-8126-00-05_2	Charge Air Tube		
1	DIN 3771 46.2x3.55 MVQ 70	O-Ring		

Material is available from Diamond Aircraft Industries.



III INSTRUCTIONS

If RSB 40NG-011 or MÄM 40-506 or MÄM 40-502 was carried out:

1	Make sure AMM-TR-MÄM 40-530 has been incorporated into the AMM.		
2	Remove cowlings (i.a.w. AMM section 71-10).		
	WARNING WARNING	DO NOT STAND WITHIN THE DANGER AREA OF THE PROPELLER. IF THE ENGINE STARTS, THE PROPELLER CAN CAUSE DEATH OR INJURY TO PERSONS. MAKE SURE THAT THE ENGINE IS SAFE BEFORE YOU WORK ON THE ENGINE CONTROLS. IF THE PROPELLER TURNS IT CAN CAUSE INJURY TO PERSONS.	
3	Remove current charge air tubing between turbo charger and intercooler:		
	 Remove a D44-8122 Remove a charge air 	and discard all used (8) worm drive clamps on the turbo connector assy 2-20-00 and the charge air tube D44-8126-00-05_1. and discard the turbo connector assy D44-8122-20-00 (including o-ring) and r tube D44-8126-00-05_1.	
4	 Install charge air tube D44-8126-00-05_2 and turbo connector assy D44-8122-20-00_2 (D44-8122-20-00_4). (refer to SI 40NG-007 latest revision) Install the Turbo Connector Assy including o-ring DIN 3771 46.2x3.55 MVQ 70. Do NOT tighten the V-Clamp at the turbo charger at this time. Install the charge air hose D44-8126-00-08 to the turbo connector assy D44-8122-20 00_2 (or D44-8122-20-00_4). Install the charge air tube D44-8126-00-05_2 to the charge air hose D44-8126-00-08 and move the 8 new worm drive clamps into position (refer to SI 40NG-007 latest revision). Connect the charge air tube D44-8126-00-05_2 to the charge air hose D44-8122-00-10_1. 		
	CAUTION	DO NOT TIGHTEN THE WORM DRIVE CLAMPS WITH THE ENGINE BEING HOT. THIS CAN DAMAGE THE SILICONE HOSES AND MAY RESULT IN LOSS OF ENGINE POWER.	











7	Tighten worm drive clamps (SI 40NG-007, latest revision):
	 Position and tighten the worm drive clamps on the charge air hoses D44-8122-00-10_1 and D44-8126-00-08 according to SI 40NG-007 latest revision.
	CAUTION DO NOT TIGHTEN THE WORM DRIVE CLAMPS WITH THE ENGINE BEING HOT. THIS CAN DAMAGE THE SILICONE HOSES AND MAY RESULT IN LOSS OF ENGINE POWER.
	• Apply torque seal on the worm drive clamps (see Figure 3 below).
	Figure 5: Worm drive clamp torque seal• Tighten the V-clamp on the turbo connector assy with 5 Nm.
8	Clean working areas, check for foreign objects.
9	Check all altered, replaced, repaired parts for proper function.
10	Test all systems in working area for function.
11	Make all necessary entries in the airplane logs.

If RSB 40NG-011 or MÄM 40-506 or MÄM 40-502 was NOT carried out:

12	Make sure AMM-TR-MÄM 40-530 has been incorporated into the AMM.		
13	Remove cowlings (i.a.w. AMM section 71-10).		
	WARNING WARNING	DO NOT STAND WITHIN THE DANGER AREA OF THE PROPELLER. IF THE ENGINE STARTS, THE PROPELLER CAN CAUSE DEATH OR INJURY TO PERSONS. MAKE SURE THAT THE ENGINE IS SAFE BEFORE YOU WORK ON THE ENGINE CONTROLS. IF THE PROPELLER TURNS IT CAN CAUSE INJURY TO PERSONS.	



14	Remove curren	t charge air tubing:
	Remove an intercooler.	d discard all used (8) worm drive clamps between turbo charger and
	Remove an	d discard the charge air elbow hose SNS5X-O-60_72-90°.
	 Remove the Remove the 	P-clamp, which holds the charged air tube on the air filter box
	(refer to Figu	ire 4 below).
		the second
		Figure 6: Attachment of charged air tube
	Remove an	d discard charge air tube D44-8126-00-05.
	Remove an	d discard charge air hose D44-8122-00-10.
15	Install charge a D44-8122-20-0 (refer to SI 40N	ir tube D44-8126-00-05_2 and turbo connector assy D44-8122-20-00_2 (or 0_4). G-007, latest revision)
) Install the h	exagenal based scrow DIN 022 M8x16 XZD with washer into the air filter bax
	where the F	-clamp was previously removed.
	 Install the T Do NOT tigl 	urbo Connector Assy including o-ring DIN 3771 46.2x3.55 MVQ 70. Inten the V-Clamp at the turbo charger at this time.
	Install the class of the c	harge air hose D44-8126-00-08 to the turbo connector assy D44-8122-20-
	 Install the cl 	narge air tube D44-8126-00-05_2 to the charge air hose D44-8126-00-08
	and move th revision).	ne 8 new worm drive clamps into position (refer to SI 40NG-007, latest
	• Connect the D44-8122-0	e charge air tube D44-8126-00-05_2 to the charge air hose 0-10_1.
	CAUTION E	OO NOT TIGHTEN THE WORM DRIVE CLAMPS WITH THE ENGINE BEING HOT. THIS CAN DAMAGE THE SILICONE HOSES AND MAY RESULT IN LOSS OF ENGINE POWER.











18	Tighten worm drive clamps (SI 40NG-007, latest revision):		
	 Position and tighten the worm drive clamps on the charge air hoses D44-8122-00-10_1 and D44-8126-00-08 according to SI 40NG-007 latest revision. 		
	CAUTION DO NOT TIGHTEN THE WORM DRIVE CLAMPS WITH THE ENGINE BEING HOT. THIS CAN DAMAGE THE SILICONE HOSES AND MAY RESULT IN LOSS OF ENGINE POWER.		
	Apply torque seal on the worm drive clamps (see Figure 11 below).		
	Figure 11: Worm drive clamp torque seal • Tighten the V-clamp on the turbo connector assy with 5 Nm.		
19	Clean working areas, check for foreign objects.		
20	Check all altered, replaced, repaired parts for proper function.		
21	Test all systems in working area for function.		
22	Make all necessary entries in the airplane logs.		
23	Make all necessary entries in the airplane logs.		